

Space Environments for Satellite Operations and Design: Overview, Mismatch, and GOES



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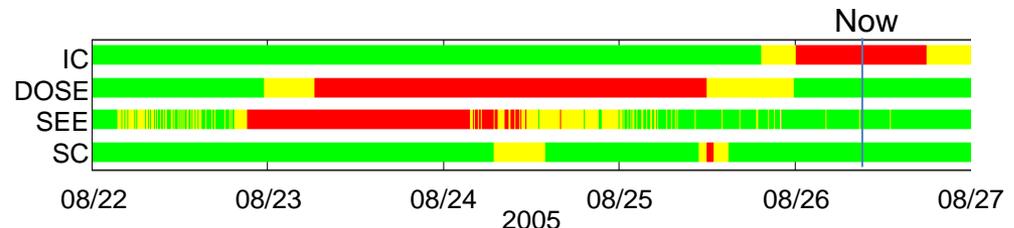
The Aerospace Corporation

Outline

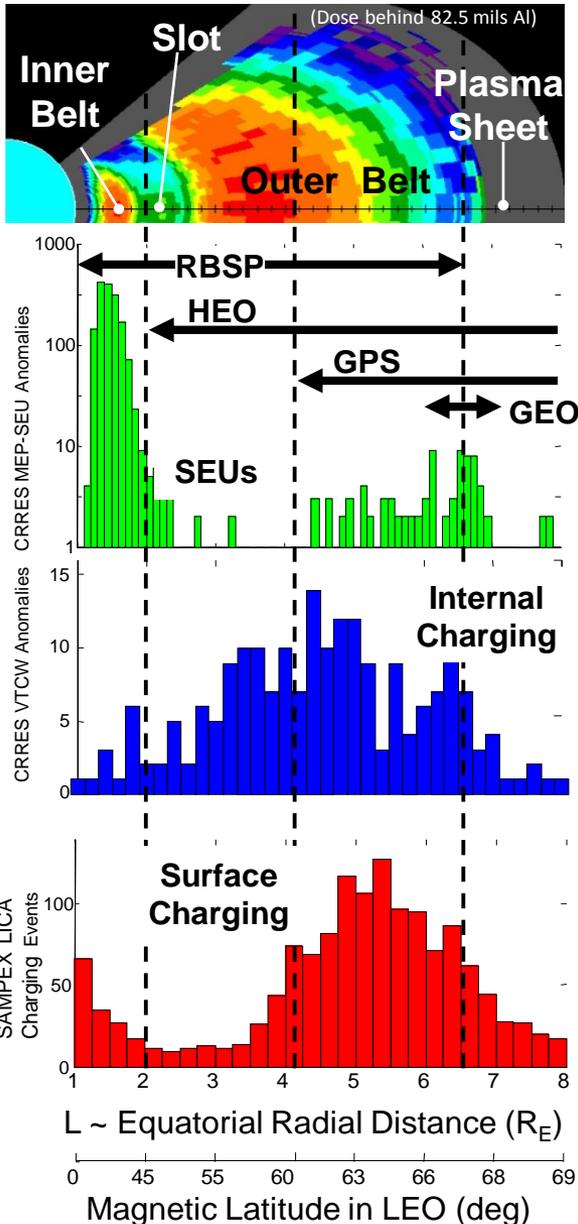
- Operations Community (Satellite Anomalies)
 - Driven by Forensics
 - Environment at the Spacecraft
 - Mission Context
 - The Mismatch
- Design Community (Design Specifications)
 - Needs Climatology
 - Mean Environments
 - Worst Case Environments
 - Probabilities
 - Space “Climate Change”
 - The Mismatch
- Summary

What Do Satellite Operators Want?

- Mostly forensics
 - Satellite anomalies are rare, even under extreme space weather conditions
 - Forecasts provide awareness, but rarely lead to action
- Many do not know what they want
 - They do not have the resources to investigate most of their anomalies
 - If they can recover the system, they may not investigate the cause
- Very few want raw physical quantities (e.g., particle fluxes)
 - Most operators cannot give you a threshold above which they have an anomaly
- Most want a simple indication of what time the hazard is greatest and how that compares to what the vehicle has already seen recently and since launch



What are the Hazards?

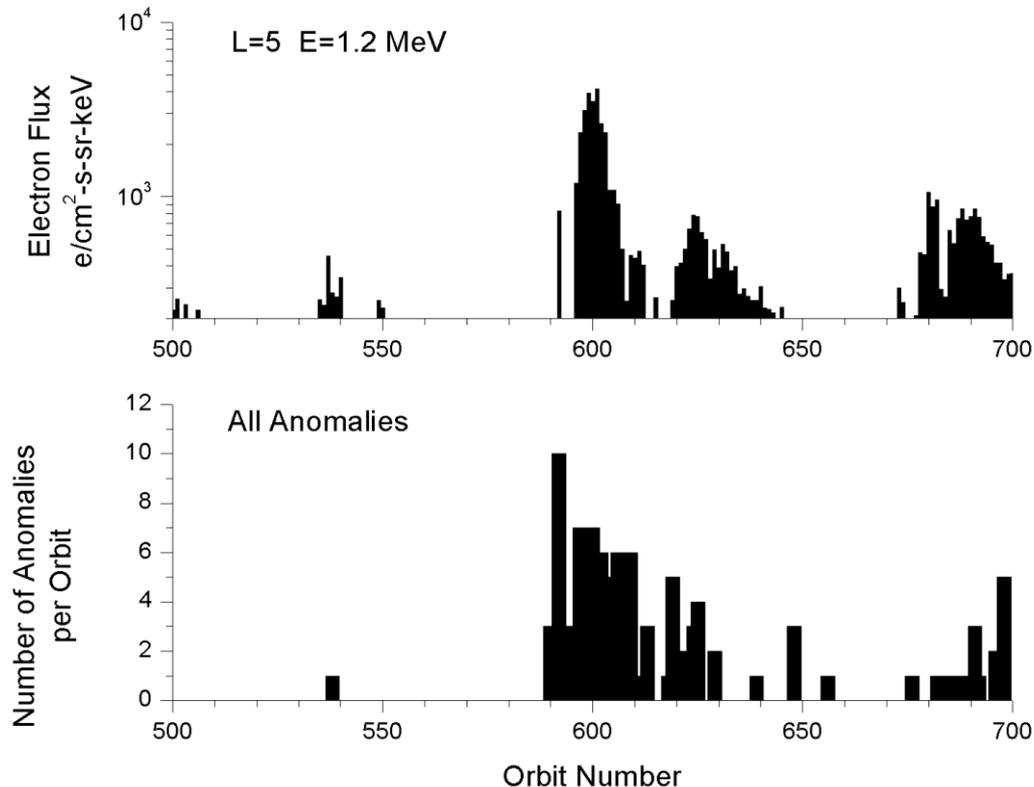


- Event Total Dose
 - Caused by \sim MeV electrons and multi-MeV protons
 - Driven by flux intensity
 - Requires hours to days of accumulation
 - Typically need evidence on same L shell or range of L shells
- Single Event Effects
 - Caused by multi-MeV protons and heavy ions
 - Driven by flux intensity
 - Instantaneous
 - Typically need evidence on same L shell or range of L shells
- Internal Charging
 - Caused by >0.1 MeV electrons
 - Driven mainly by flux intensity
 - Requires hours of accumulation
 - Typically need evidence on same L shell or range of L shells
- Surface Charging
 - Caused by keV electrons
 - Driven mainly by spectral shape, not flux intensity
 - Usually diagnosed with L , MLT or local temperature/spectrum
 - Timing and location are *hugely* critical

Location, Location, Location

- The biggest challenge today supporting satellite operators is that they do not have environmental sensors on their vehicles
 - The environment at GOES and POES are not representative of all of geospace
 - Without data, we are left guessing the environment
- Tools for “filling in” the environment everywhere should be a high priority
 - Dynamic geomagnetic cutoff models coupled to L1/GEO fluxes
 - Data assimilation
 - Statistical models driven by data available with low latency (24 hrs)
- Tools to project the 3-D environment onto a satellite orbit

Mission Context



CRRES Example

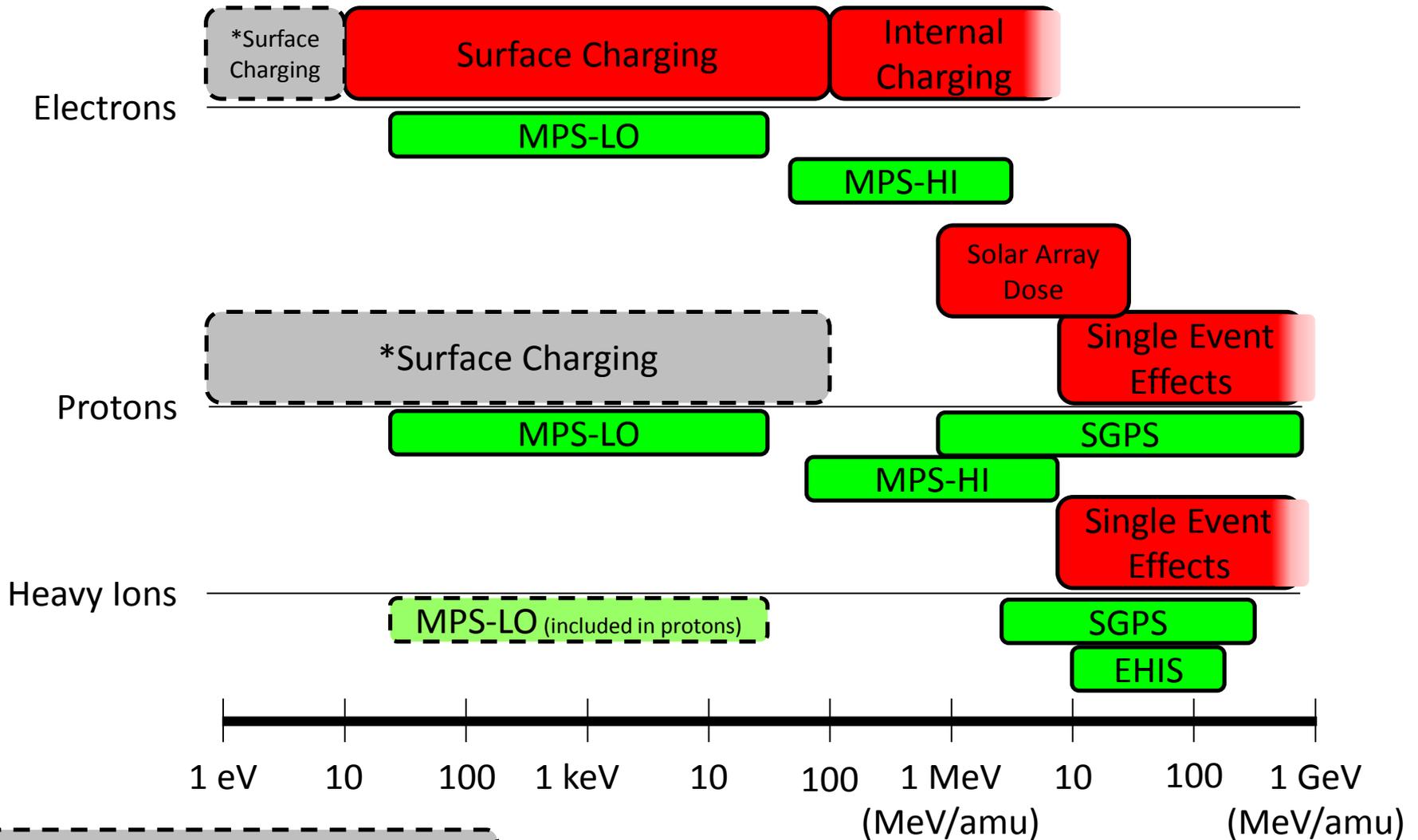
- The operator should ask: is the environment at the time of the anomaly worse than other times when there were no anomalies?
- To answer this, we need to know the entire history of the vehicle's environment back to launch

The Magic Numbers

- Provide these top-priority quantities:
 - FAC intensity (LEO) [Surface Charging]
 - Kp [Surface Charging]
 - Electron temperature [Surface Charging]
 - 0.1-2 MeV electron flux [Internal Charging]
 - 20-50 MeV proton flux (heavy ions would be nice, too) [SEE]
 - 1-20 MeV proton flux [Event Total Dose]
- At the operator's spacecraft from near term forecast all the way back to launch
- For forensics, the immediate past (~24 hrs) is the highest priority



Operational Hazards and GOES-R Measurements



*Provides Supplemental Info

The Mismatch: Operations

- The scientific community is not working on all of these things
 - Until recently electrons were omitted from ring current models
 - Scientists are learning to run MHD models in real time, but not solar particle cutoff models
 - There is little work on statistical models to fill in the 3-D environment on operational timescales
- Why?
 - Most science funding supports curiosity driven research, geared towards understanding the environment
 - The operator community does not typically fund scientists
 - Communication between the scientific and operator communities is very limited:
 - Operators do not attend science meetings or contribute to steering committees
 - Operators cannot always talk openly about their needs

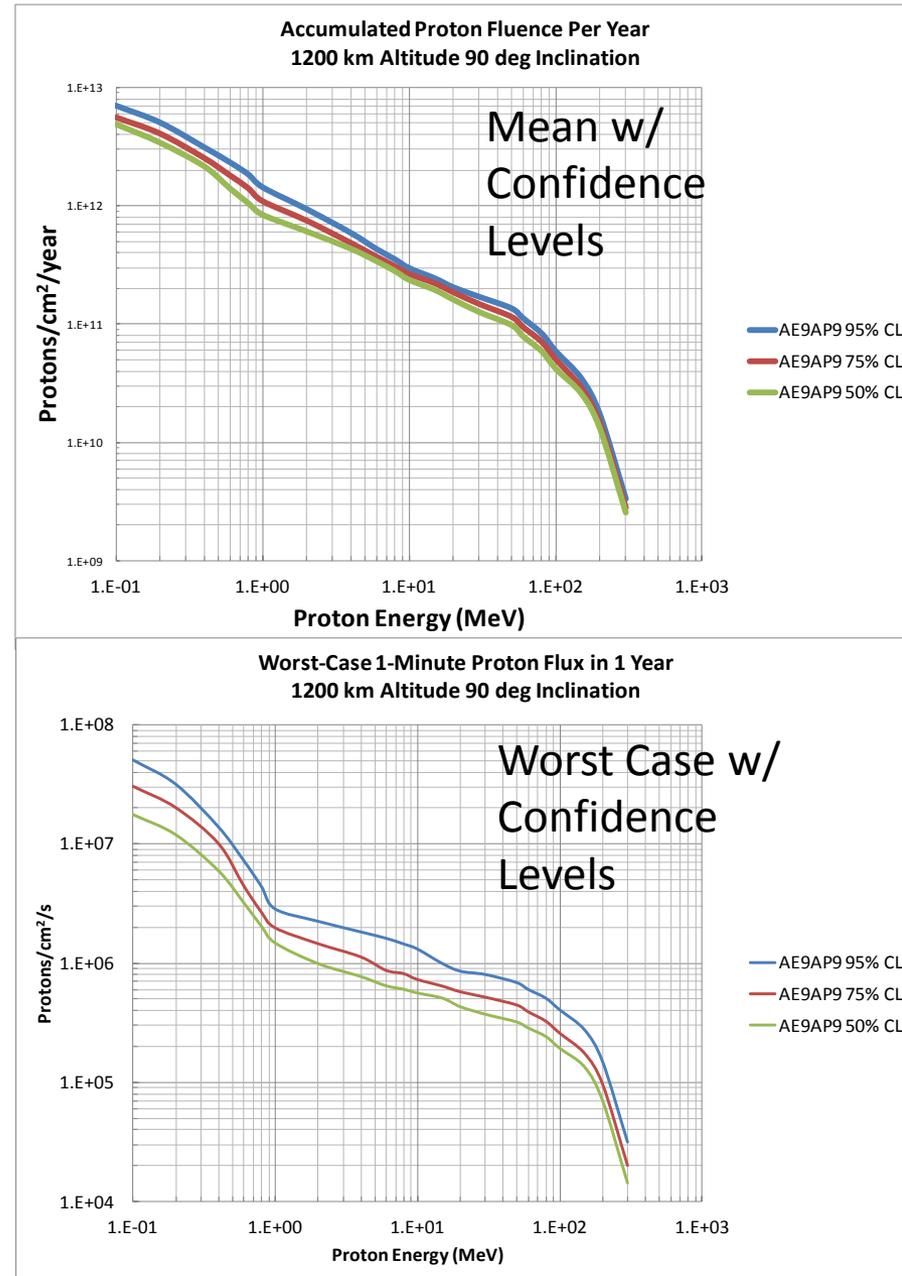
What do Satellite Designers Want?

Priority	Species	Energy	Location	Sample Period	Effects
1	Protons	>10 MeV (> 80 MeV)	LEO & MEO	Mission	Dose, SEE, DD, nuclear activation
2	Electrons	> 1 MeV	LEO, MEO & GEO	5 min, 1 hr, 1 day, 1 week, & mission	Dose, internal charging
3	Plasma	30 eV – 100 keV (30 eV – 5 keV)	LEO, MEO & GEO	5 min, 1 hr, 1 day, 1 week, & mission	Surface charging & dose
4	Electrons	100 keV – 1 MeV	MEO & GEO	5 min, 1 hr, 1 day, 1 week, & mission	Internal charging, dose
5	Protons	1 MeV – 10 MeV (5 – 10 MeV)	LEO, MEO & GEO	Mission	Dose (e.g. solar cells)

- NASA and the AE9/AP9 team did extensive outreach to obtain this list of requirements
- Notice that it covers everything, everywhere on all timescales
- AE9/AP9 covers this, except Plasma on sub-mission timescales

Design Specifications

- Design Specifications are derived from Climatology models
 - Mean environment for arbitrary orbit
 - Worst case environment for arbitrary orbit
 - Confidence intervals to help trade design vs risk
- AE9/AP9 (IRENE) addresses much of this, except surface charging
- Extreme environments *without* probability of occurrence are very hard to use (e.g., superstorms)



Space “Climate Change”

- Will the next solar cycle resemble the last ones, at least statistically?
- This is a critical assumption in the use of all climatology models
- What does it mean for the satellite designer if the next solar cycle is “big” or “small” in sunspot number?

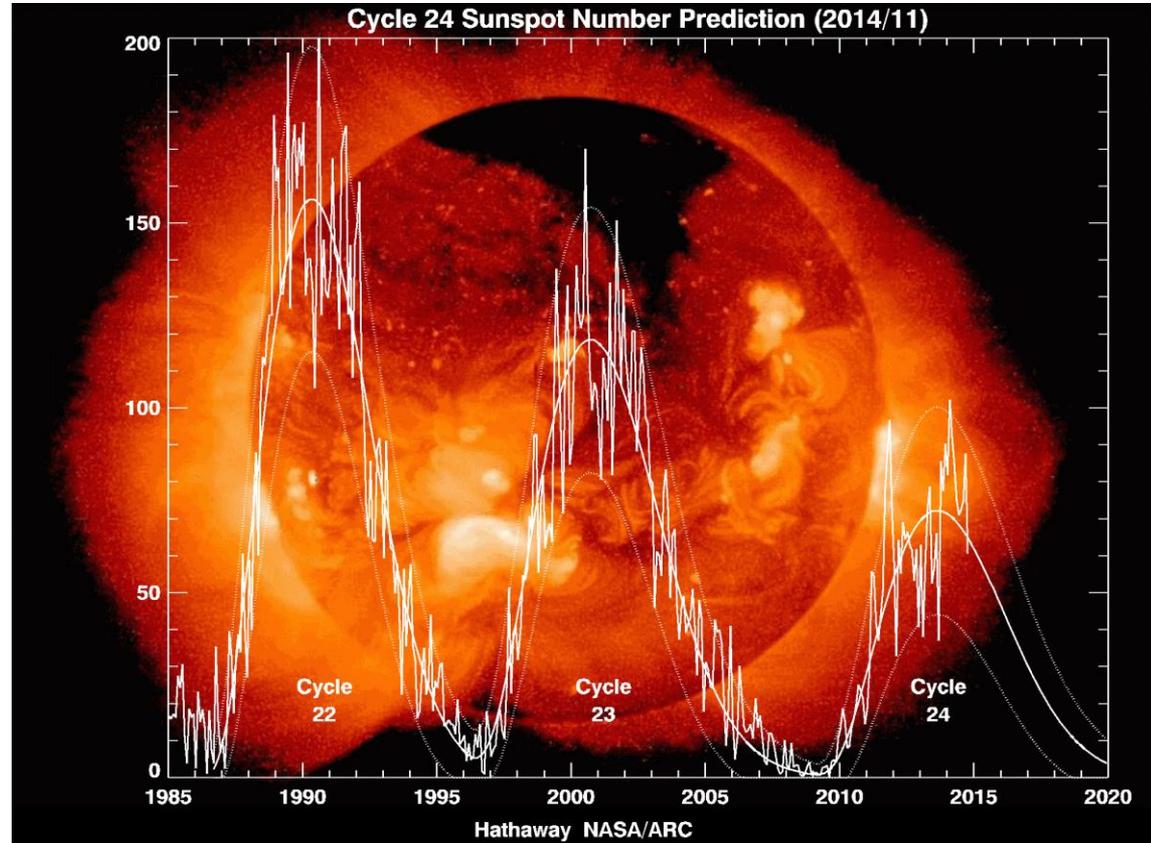


Figure courtesy of NASA/MSFC

The Mismatch: Design

- Still, there is work to do:
 - Extreme events need context:
 - How likely are they over the next couple decades?
 - What are the uncertainties in the extrapolations?
 - Space “Climate Change”: Will the next solar cycle match past climatology?
 - Where is the South Atlantic Anomaly going to be in 20 years?
 - Reanalysis Climatology
- The mismatch is not as bad as it is for operations
 - Conscious efforts at outreach by NASA and the AE9/AP9 team
 - Some separate funding: AE9/AP9
 - Satellite design concerns tend to be less *sensitive*, so communication is easier

GOES Next Measurements

- Particles are a top priority for both anomaly resolution and satellite design
 - The primary hazards to satellites are caused by plasma and radiation particles
 - GOES-R adds energy resolution that permits maximum correlation between environmental measurements and satellite anomalies
- Magnetometer plays a critical supporting role:
 - Detecting magnetic reconfigurations
 - Mapping particle fluxes to other locations
 - Deeper scientific investigations (storm dynamics, wave-particle interactions)
- Example new measurement to consider: GPS occultation to detect plasmopause
 - Plasmopause marks boundary where surface charging is suppressed
 - Plasmopause marks boundary between different physical processes that modify global radiation environment

Conclusion

- The scientific community is doing relatively well supporting the satellite design community, though we could do better
 - Scientific curiosity and understanding happen to match up better
 - There is a focused, funded effort to identify and meet user needs
- We are not doing very well on satellite operations
 - Scientific curiosity and understanding do not match up well with operator needs
 - Funding is narrow or nonexistent: Can/should NOAA solve this?
 - Operators sometimes cannot tell us what they need
 - Don't know
 - Can't say